

**DATE RELEASED**

01-12-2023

DATE EFFECTIVE

01-12-2023

SUBJECT

Elevator Idler Fatigue Cracks

EFFECTIVITY**MODEL-** Ranger R7**SERIAL NO:** 10131-10227

Rev A	-INITIAL RELEASE
REV B	-SERIAL NO. Effectivity Revised -METHOD OF COMPLIANCE updated to show corrective action with upgraded design

REASON

Fatigue cracks have been reported in the field on the bracketry supporting the elevator pushrod idler assembly. This is due to high loads from the elevator pushrod fatiguing the bracket through cyclic loading. It is also believed that corrosion of the idler also contributes to this issue. If a crack continues to propagate without proper attention, this could lead to slop in the elevator control system and possibly failure of the supporting bracketry.

REQUIRED ACTION

Inspect bracket and fuselage ribs for cracks. Also note on condition of idler assembly.

TYPE OF MAINTENANCE

INSPECTION: Low (may require borescope); REPLACEMENT: Moderate

MINIMUM LEVEL OF CERTIFICATION

INSPECTION: Owner (may require borescope); REPLACEMENT: A&P

TIME OF COMPLIANCE

Inspect within the next 10hrs of flight time, then in 25 hr intervals until 100hrs has been reached. After this, inspection interval can default to annual inspections.

METHOD OF COMPLIANCE

Inspect the critical flanges as shown in Figure 2 and Figure 3. If any cracks are present, replace parts with upgraded design.

This service bulletin is null when the upgraded idler design is installed.

LABOR REQUIRED:

4hrs (if replacement required)



INSPECTION PROCEDURE

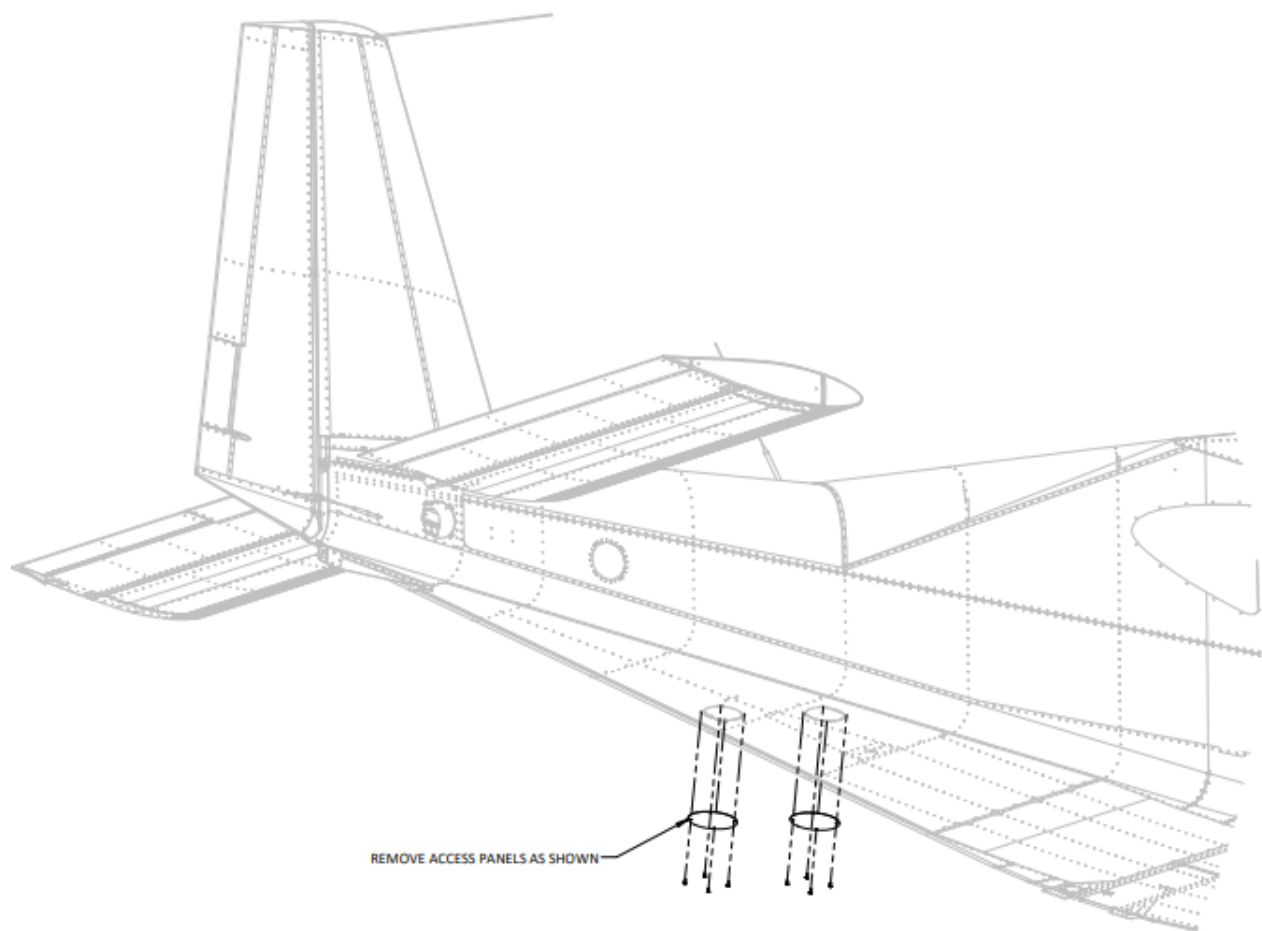


Figure 1

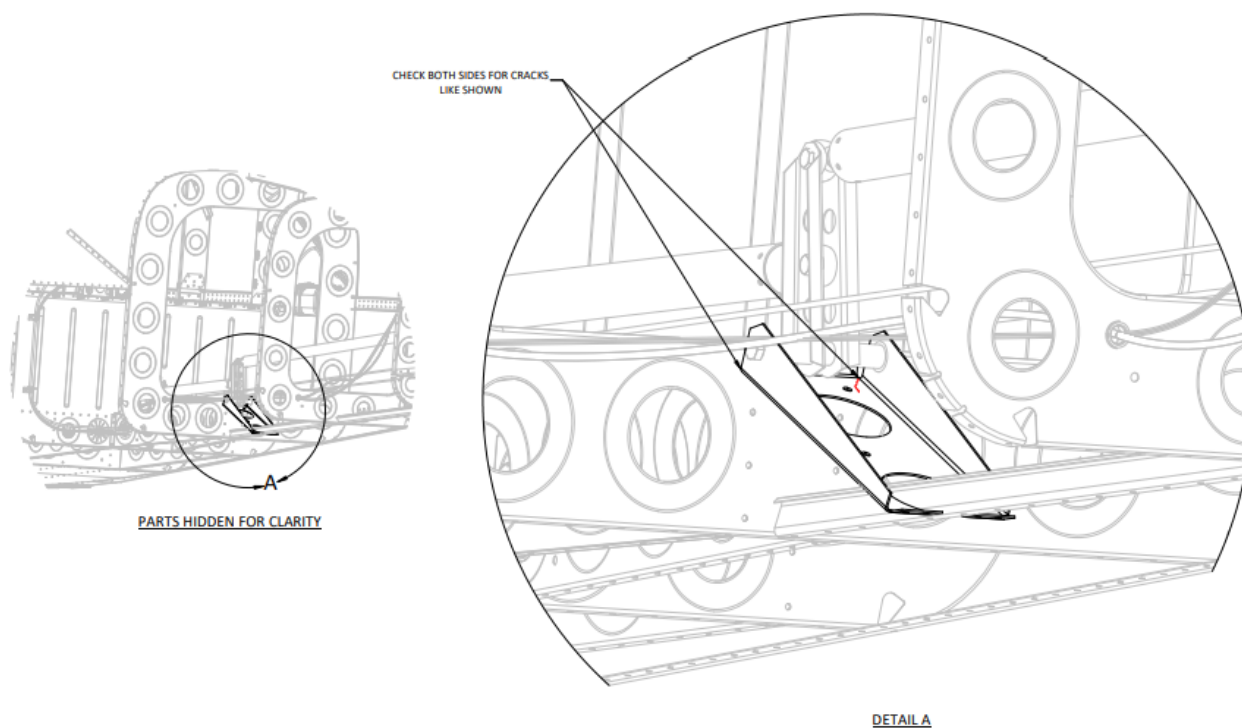


Figure 2

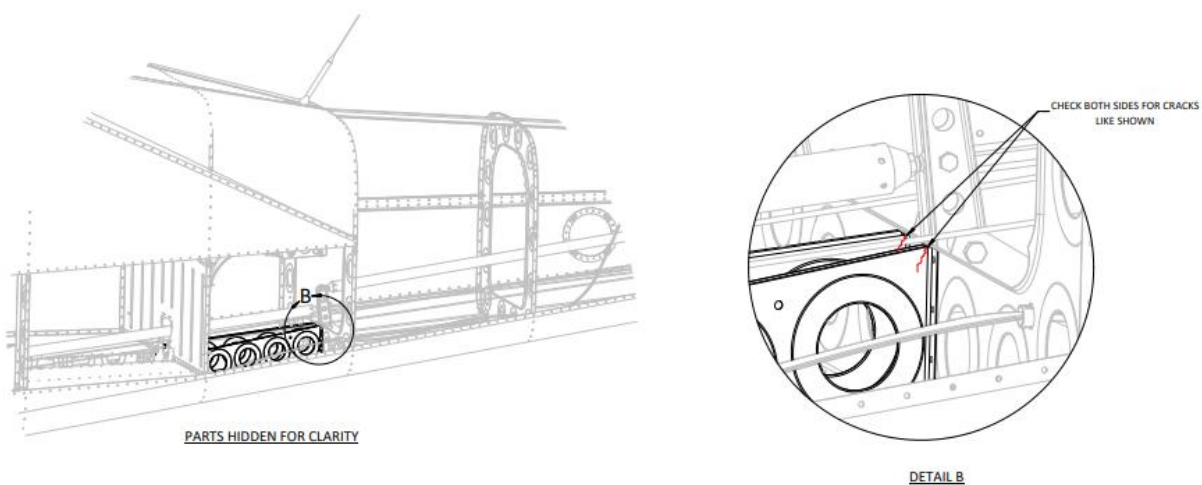


Figure 3

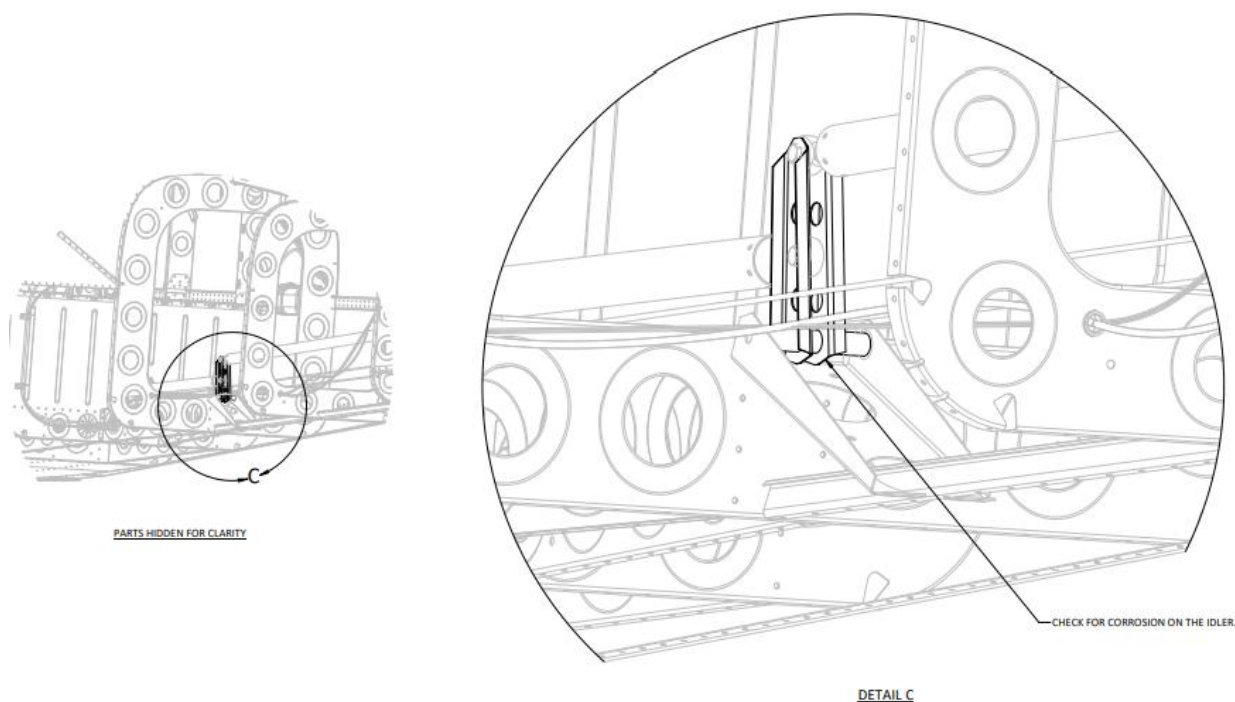


Figure 4

REPLACEMENT PROCEDURE

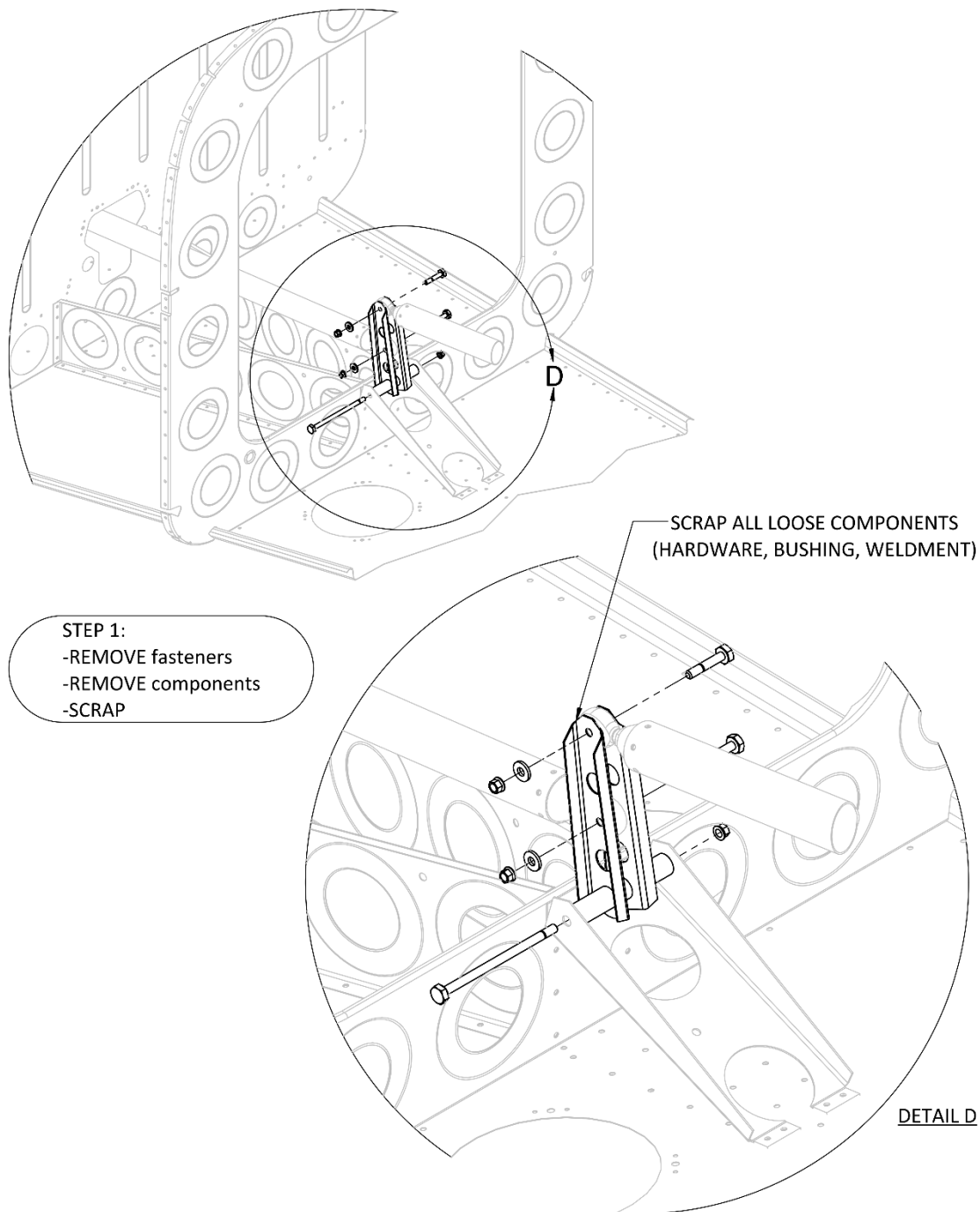


Figure 5

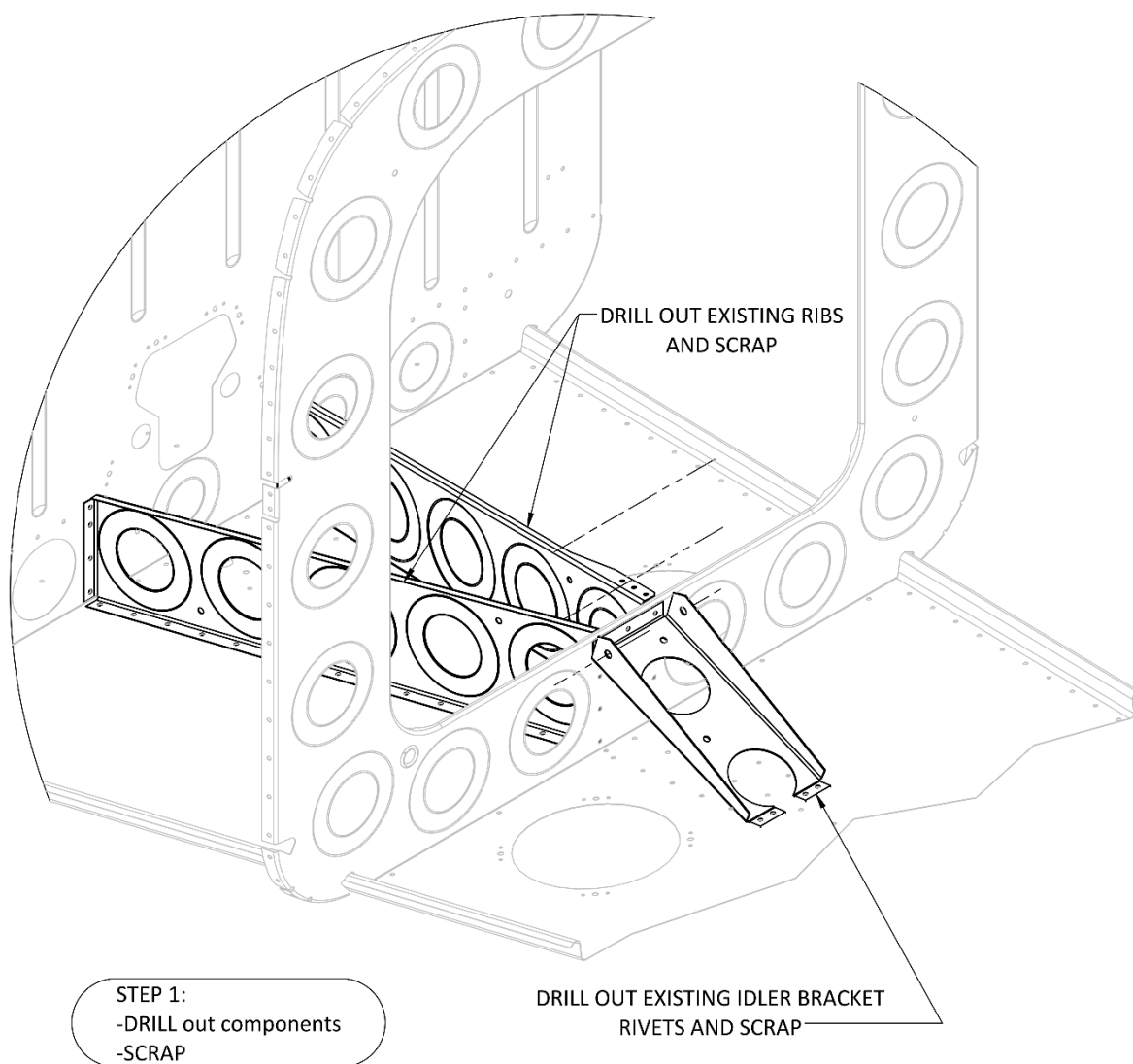


Figure 6

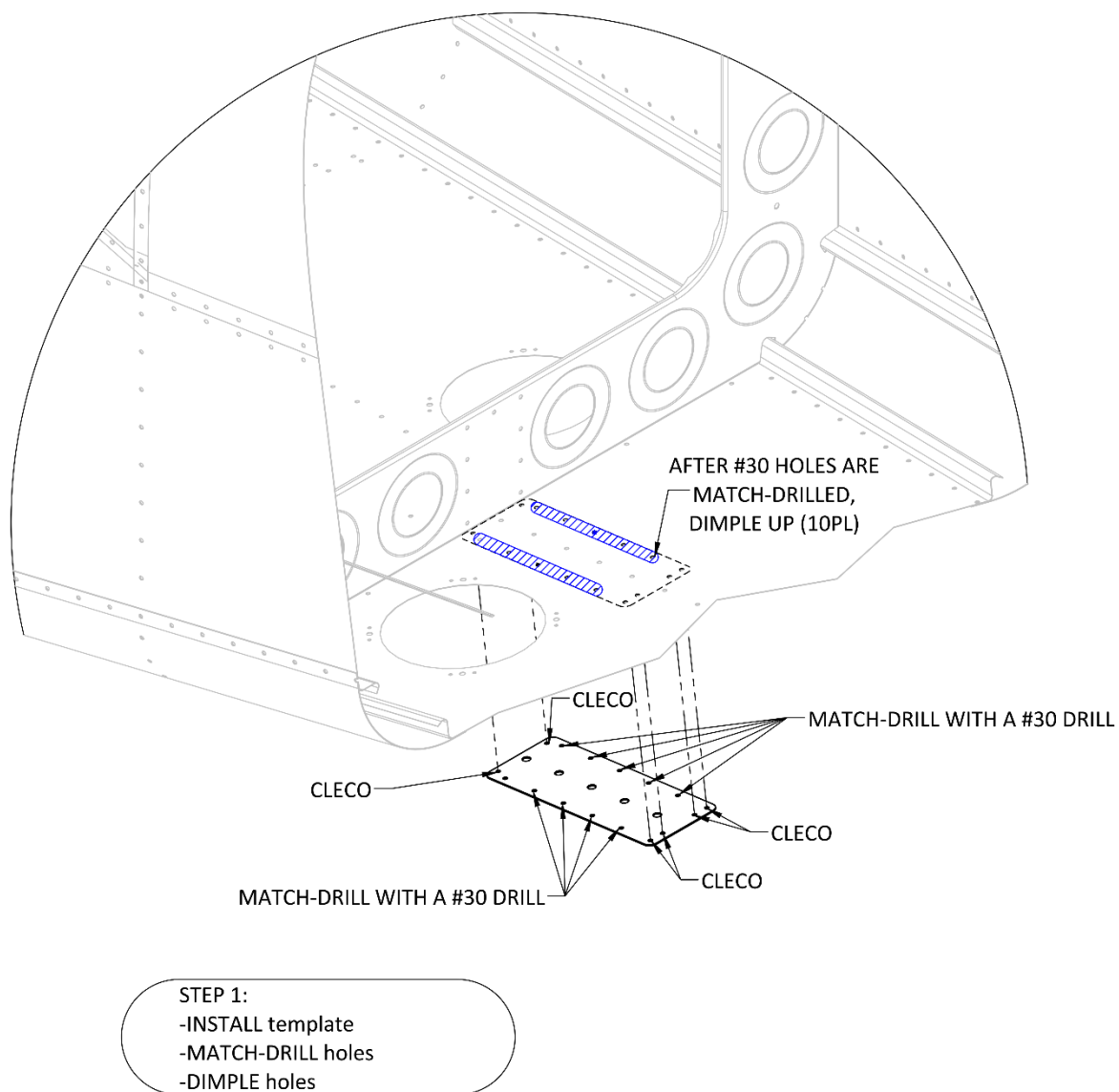


Figure 7

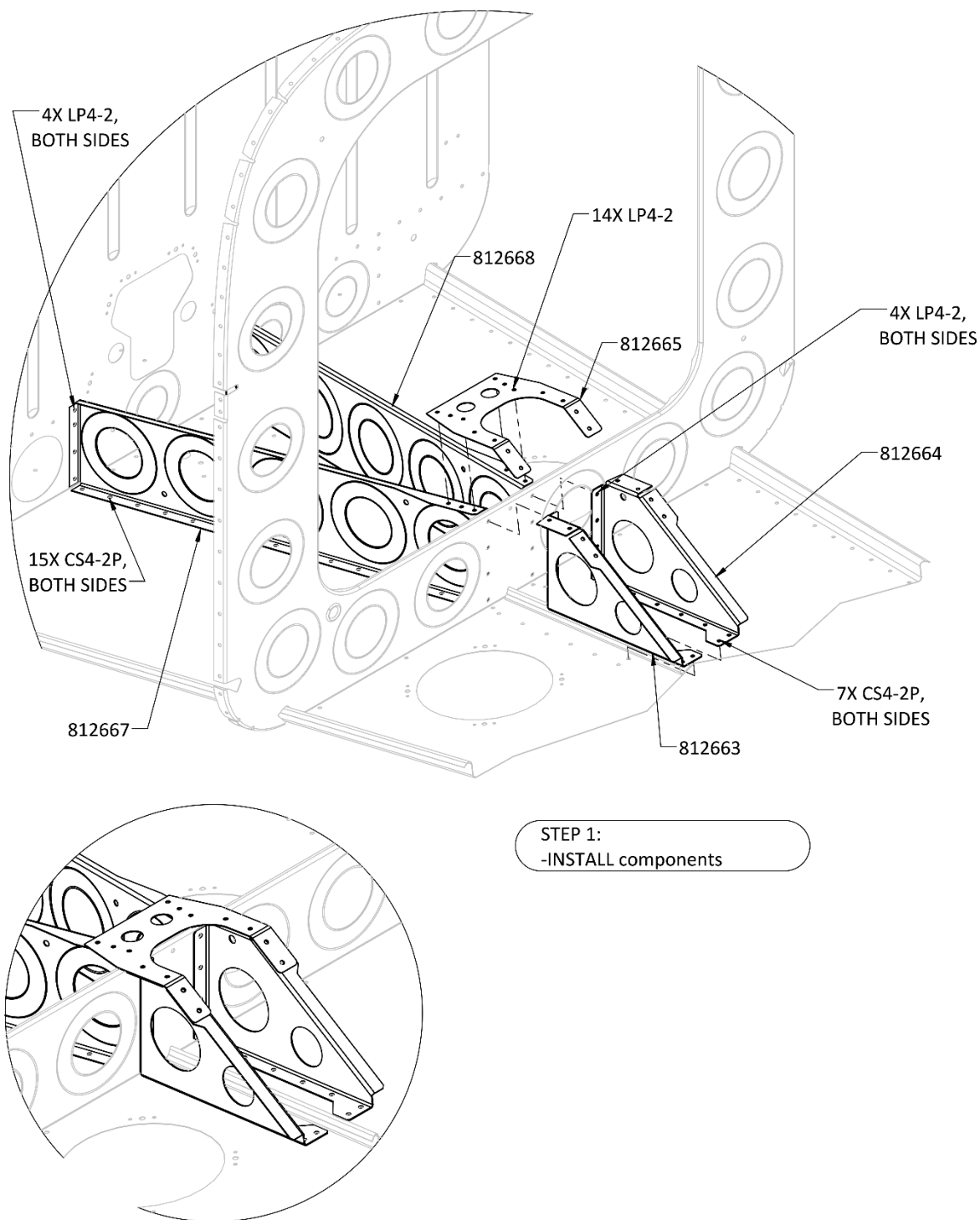


Figure 8

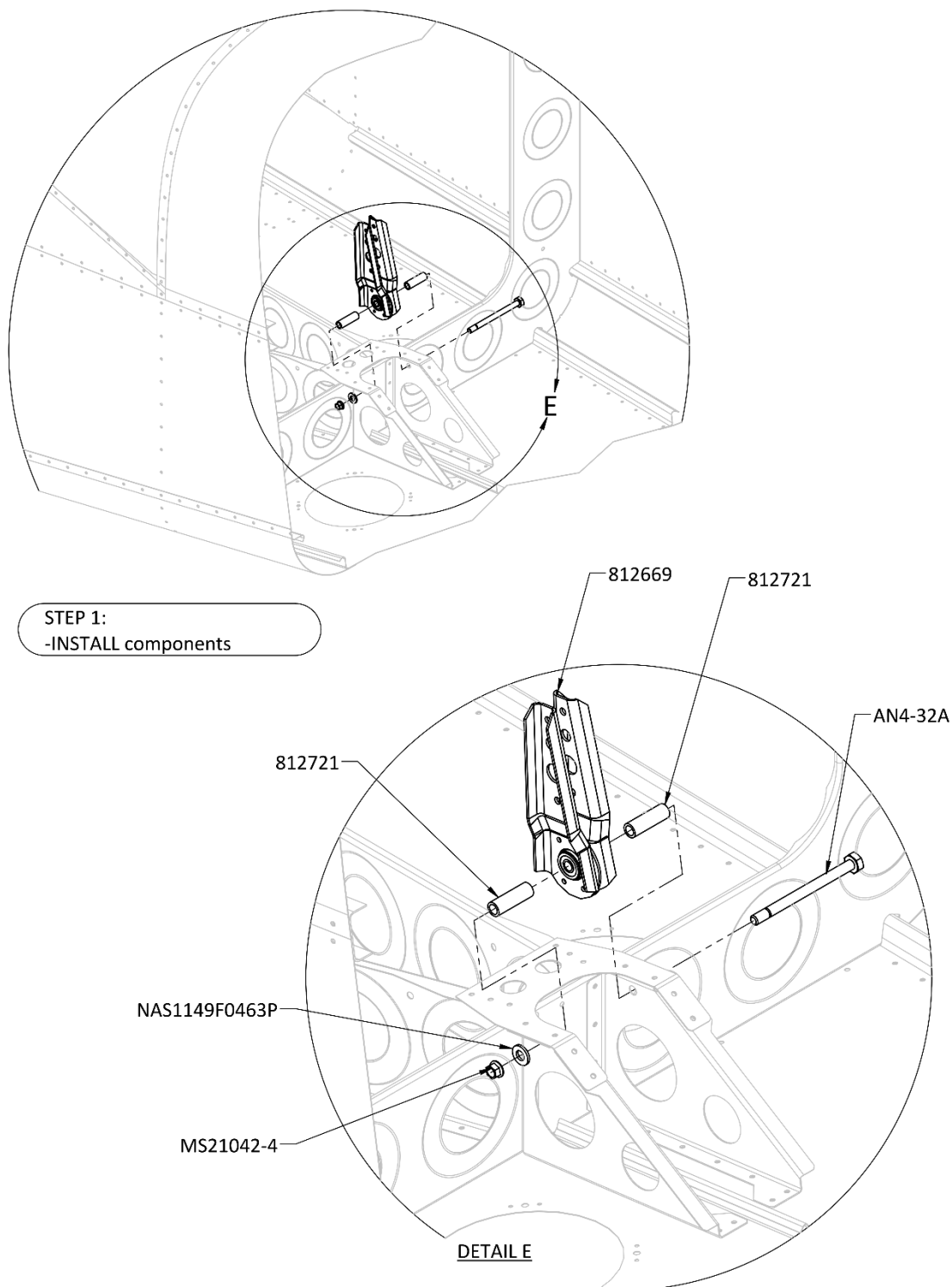
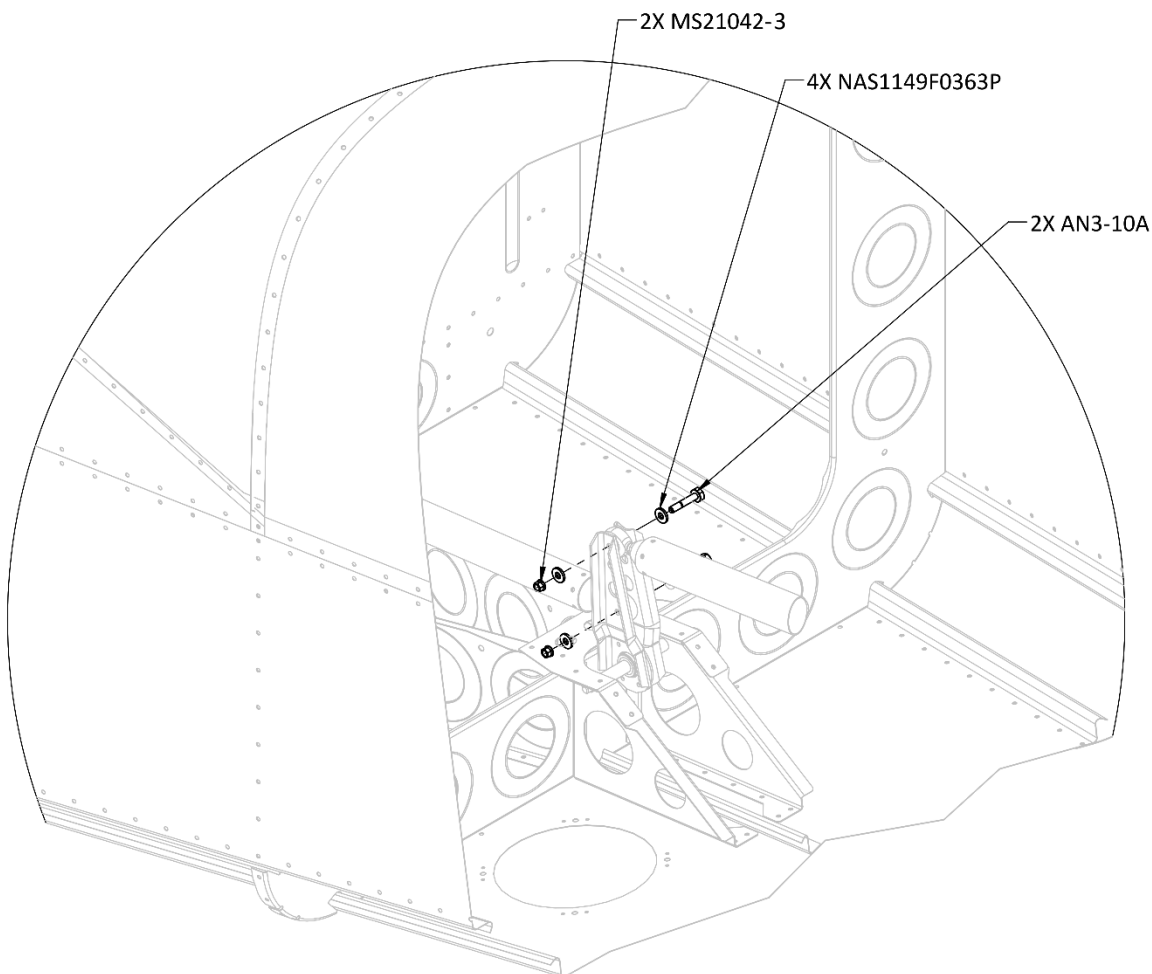


Figure 9



STEP 1:
-INSTALL components

Figure 10